FMI Acoustics Presentation





AIRCRAFT ENGINE NOISE

ROME AERONAUTICAL ENGINEERING UNIVERSITY

May the 28th, 2004

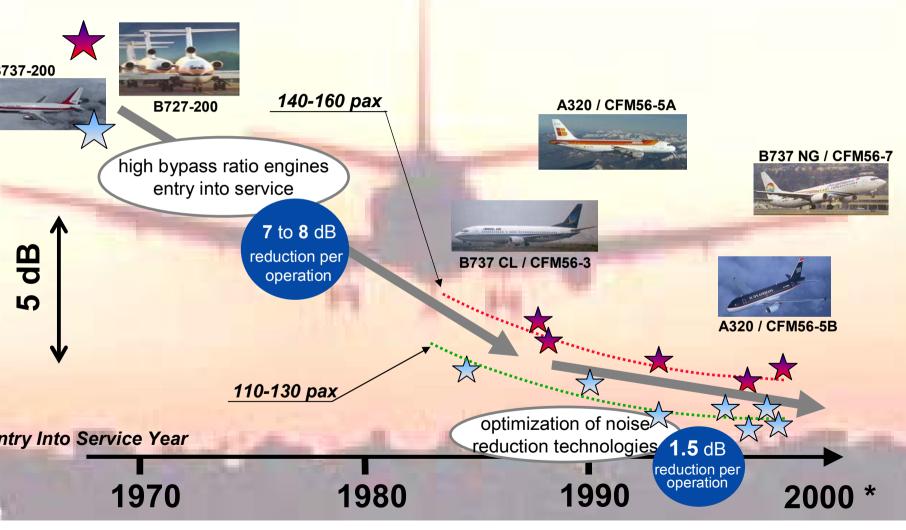
loise Reduction Trends

verage Noise Reduction in Decibels per Operation





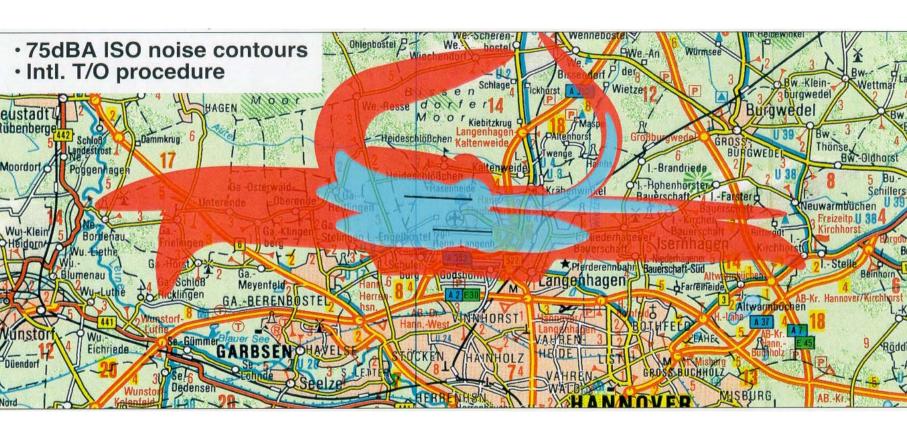
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THE POW OF FLIG

Noise Footprint Example



AME AIRCRAFT: GENERATION 1 ENGINES vs NEW GENERATION ENGINES









- TO CEMI NOISE SITUATION
- **NOISE INSIDE A NEW ENGINE PROJECT**
- **NOISE CERTIFICATION**
- TODAY'S NOISE TECHNOLOGIES
- FOR A QUIET FUTURE
- **QUESTIONS**





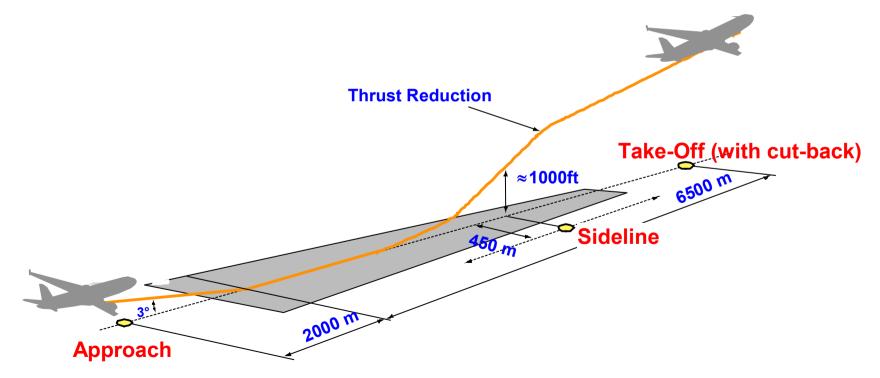
oise Certification Scheme

TATION ENVIRONMENTAL PROTECTION **ICAO ANNEX 16** Vol.1 Chapter 3 (THIRD EDITION 1993)

INT AVIATION REQUIREMENTS: **JAR 36** (1997)

DERAL AVIATION REGULATIONS : **FAR 36** (1993)

SSIAN AVIATION REGISTRATION : $\mathbf{AP-36}$ (AVIATION REGULATION - 36)

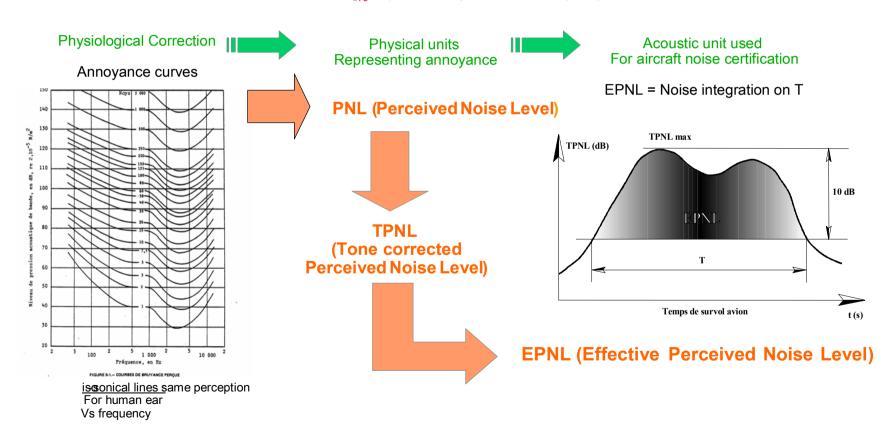






PNL: the noise aeronautical unit

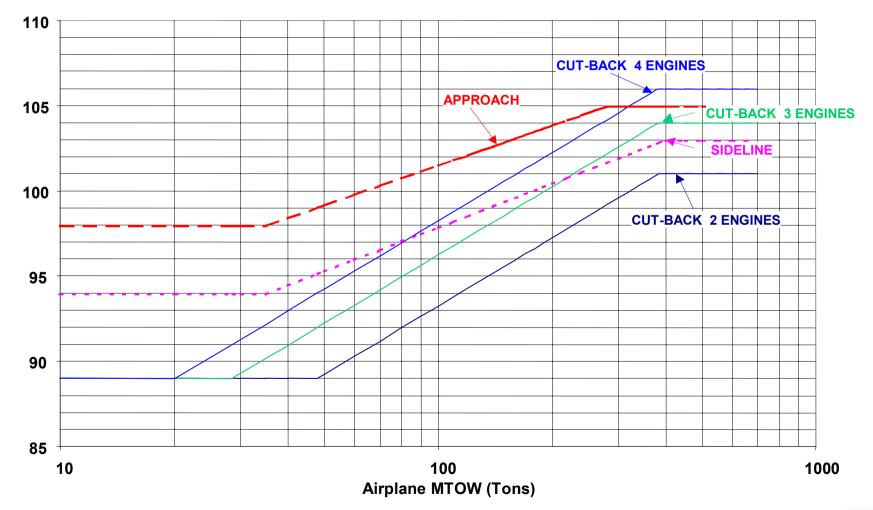
Measure Physical Unit : the décibel (dB) $dB = 10 \log_{10}$ (acoustic pressure / $20 \mu Pa$)







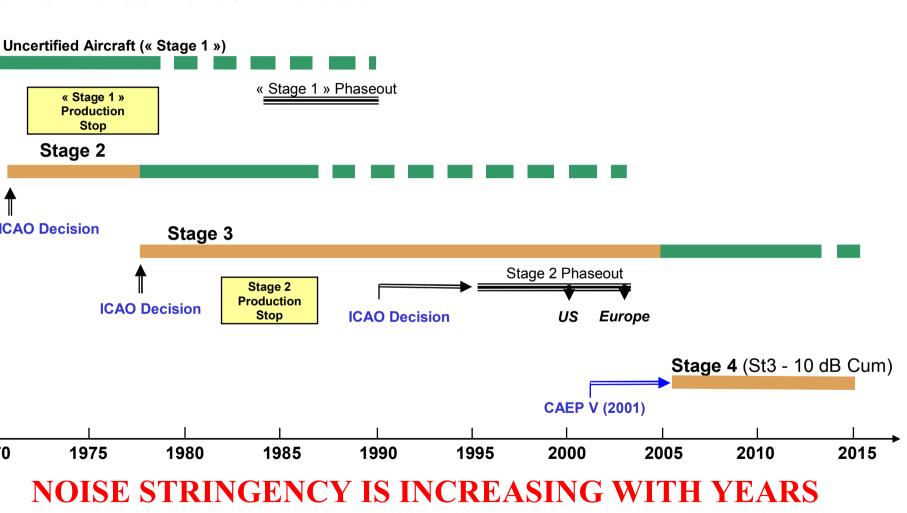
CAO Acoustic Certification- Stage 3 Limits







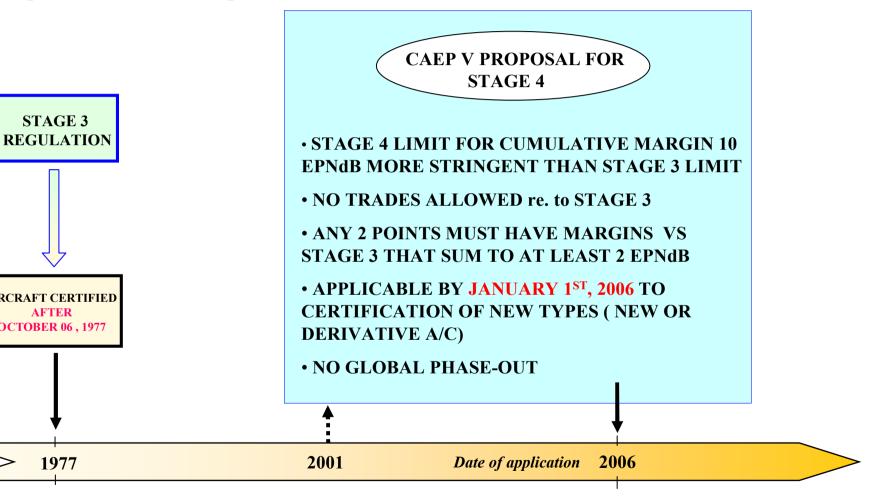
CAO Noise Rules Evolution







age 4 Noise Regulation



















FOR A QUIET FUTURE









CFM International is a Joint Company of Snecma Moteurs, France and General Electric Co., U.S.A.





FMI Noise Situation











Snecma Moteurs has prime responsibilty for Noise Engineering on all CFM56 Programmes













156-5A & 5B



156-5C

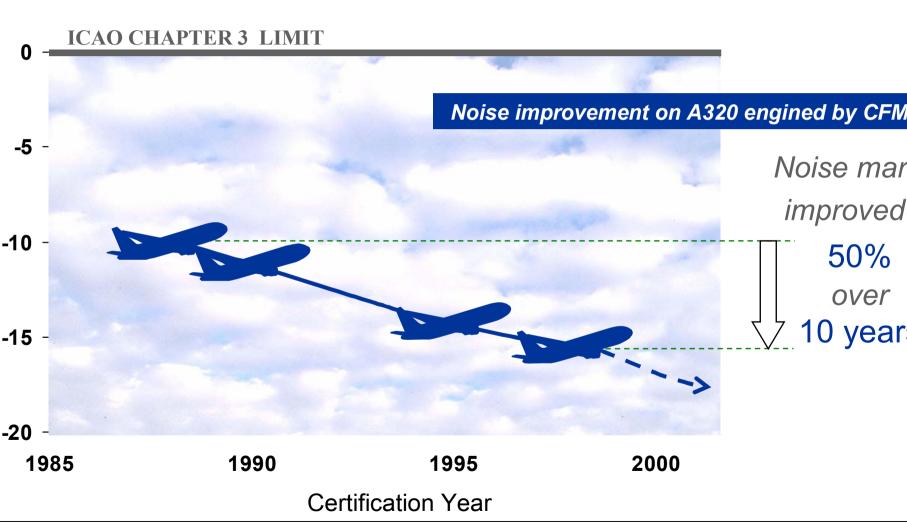


FMI Noise Situation





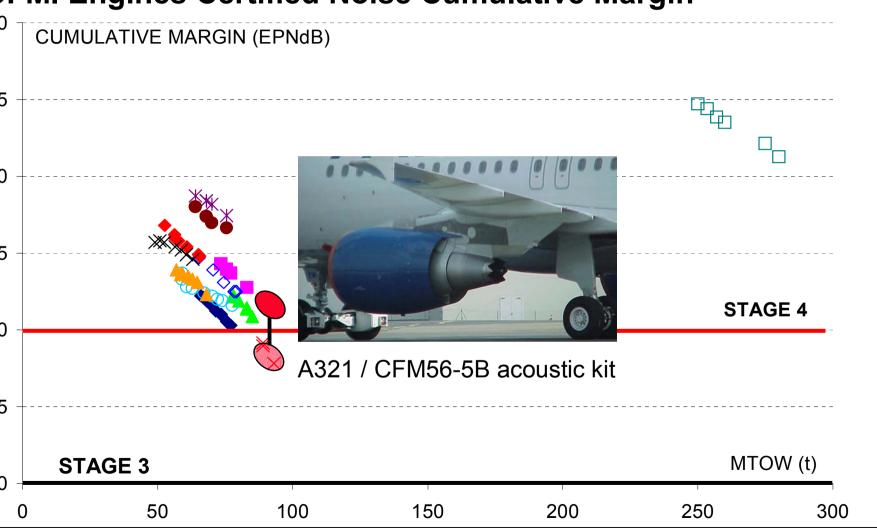
Continued Effort To Reduce Noise



FMI Noise Situation







CFMI Proprietary Information - Unauthorized Disclosure, Use, or Export are Prohibited.

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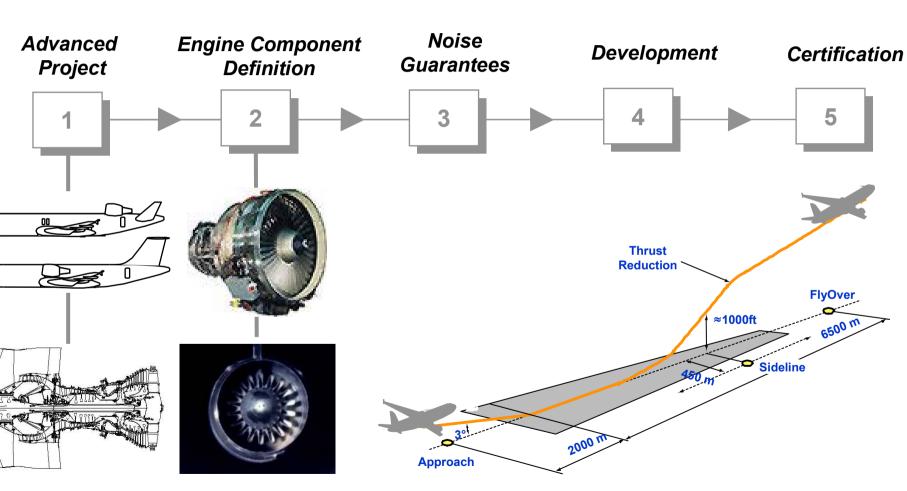








trong Integration in Engine and Aircraft Development Processe

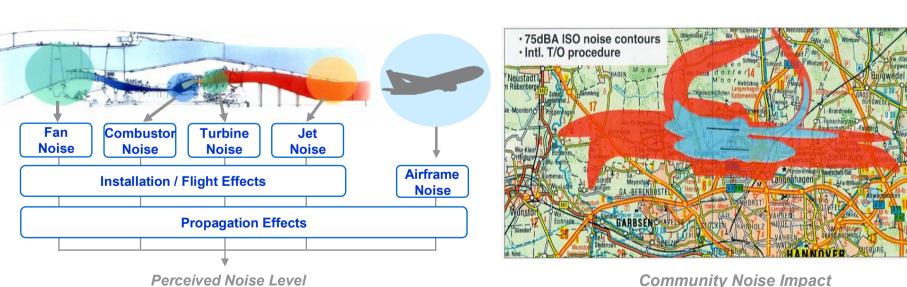






pecific Prediction Tools to support all Engine Programme Step









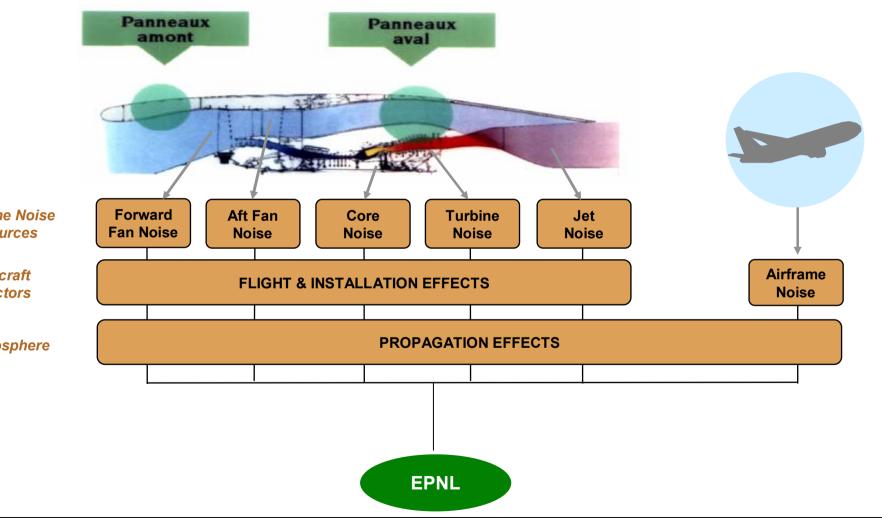
oise Sources & EPNL Calculation

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ctors

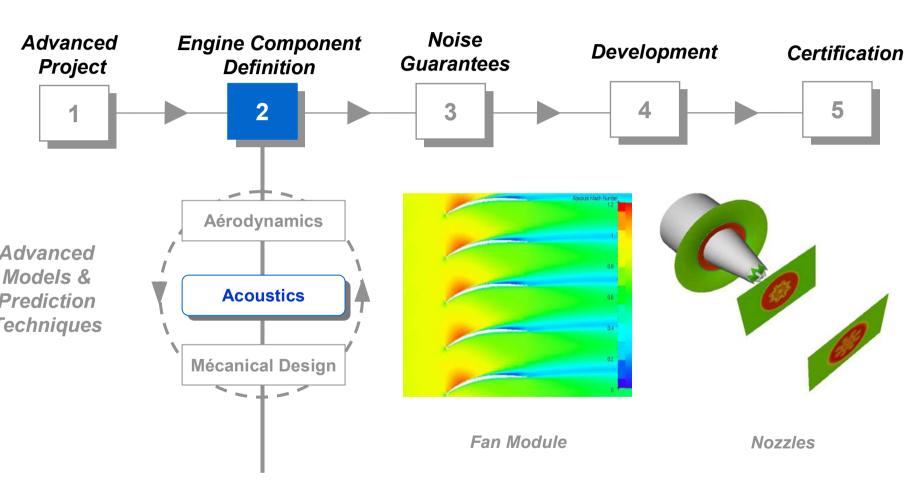
sphere







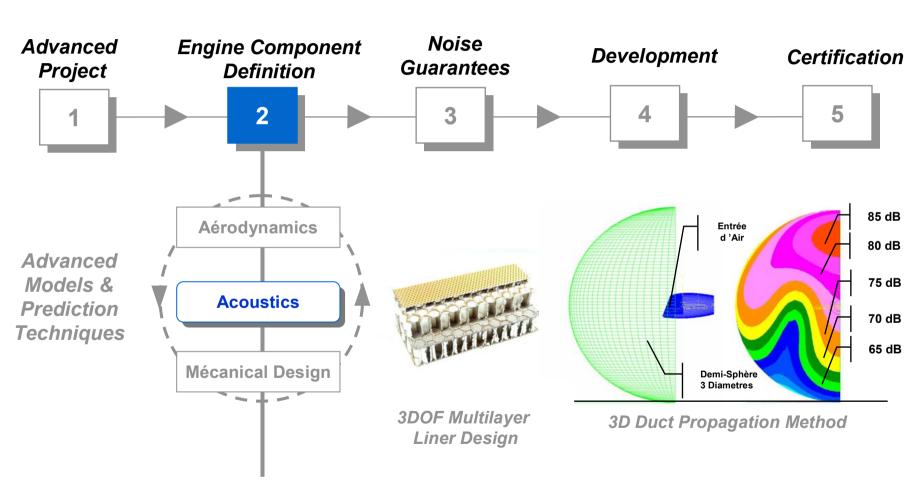
ow Noise Engine Component Design







ptimisation of Noise Reduction Systems

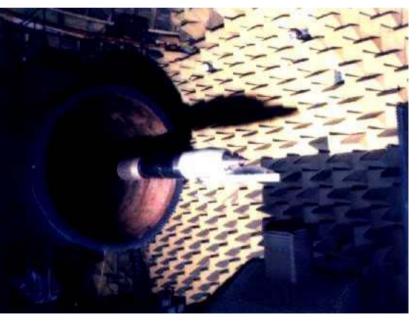






xperimental Validation









Outdoor Engine Test Facility

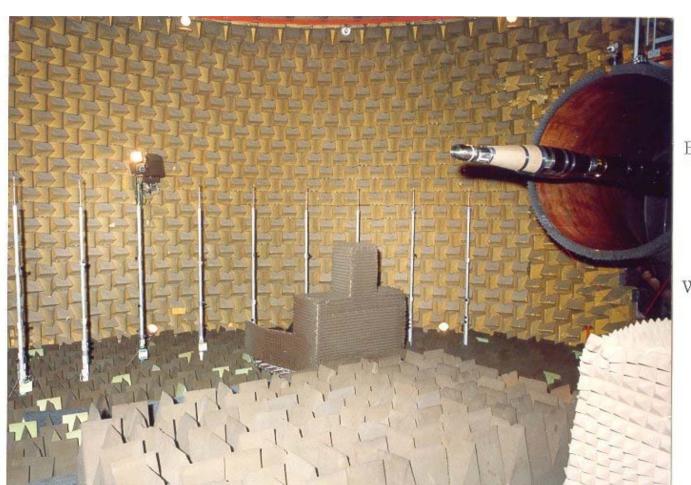




xperimental Validation

nechoic wedges

ed microphones



Exhaust nozz

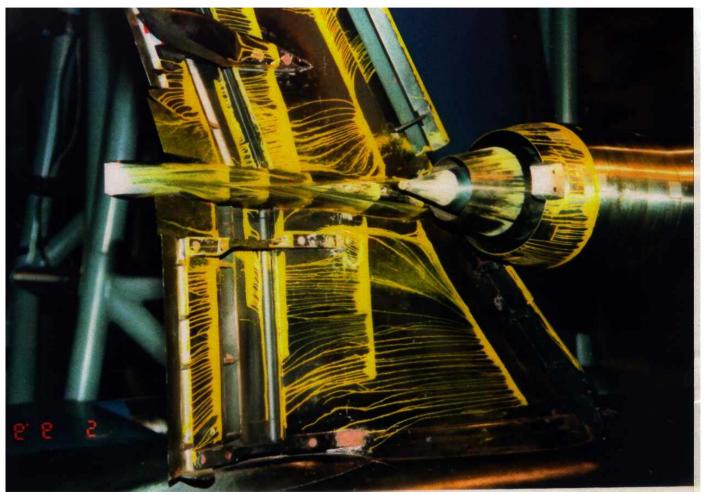
Wind Tunnel:

 $(\Phi = 2 \text{ m})$





xperimental Validation







FMI Noise Experience / Experimental Database

Jet noise model in wind tunnel

12 campaigns since 1985, more than 60 configurations

Engine static test

- more than 15 static engine acoustic certification
- more than 20 campaigns
- more than 120 configurations including acoustic liners

Flight test

- 9 engineering flight tests
- 6 certification flight tests









- TO CENTER STRUCTURE STRUCTURE TO SERVICE STRUCTURE S
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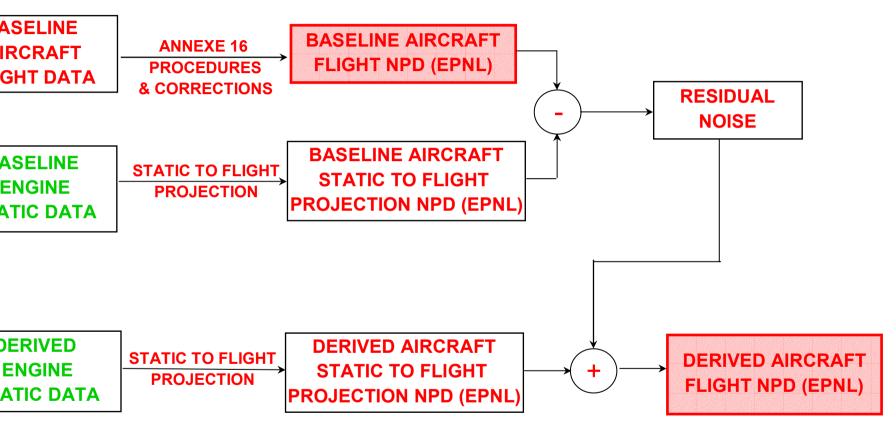
loise Certification





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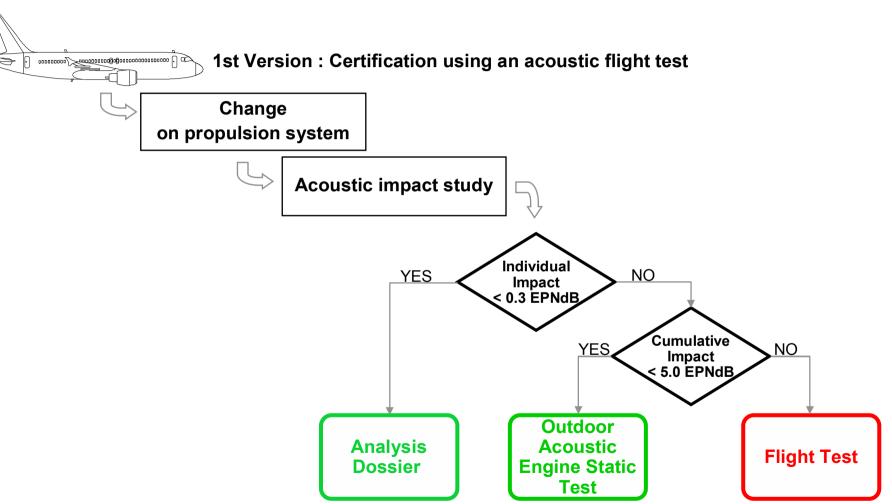
oise Family Plan Concept



loise Certification







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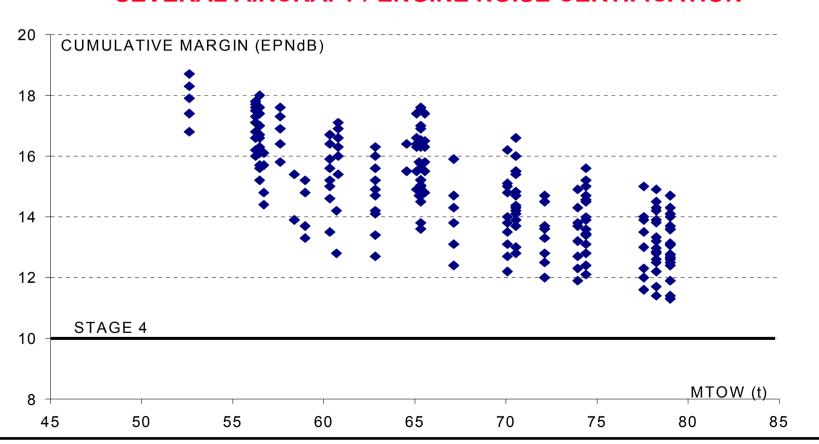
loise Certification





3737NG / CFM56 7B Example

2 CERTIFICATION STATIC TEST + 1 CERTIFICATION FLIGHT TEST = SEVERAL AIRCRAFT / ENGINE NOISE CERTIFICATION



















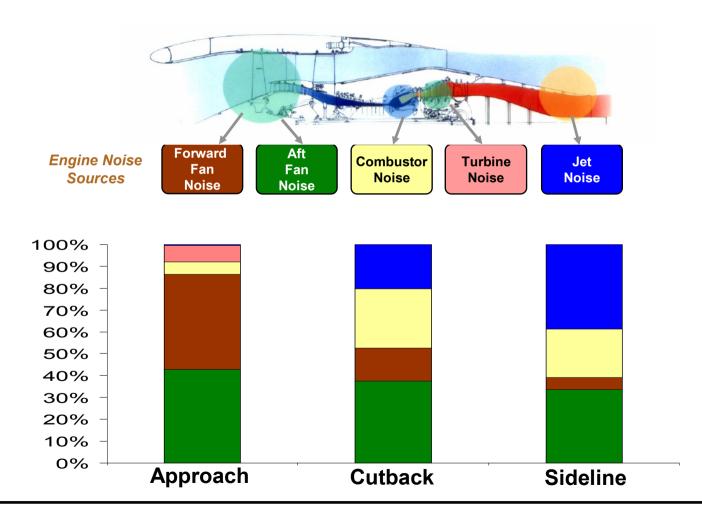
FOR A QUIET FUTURE







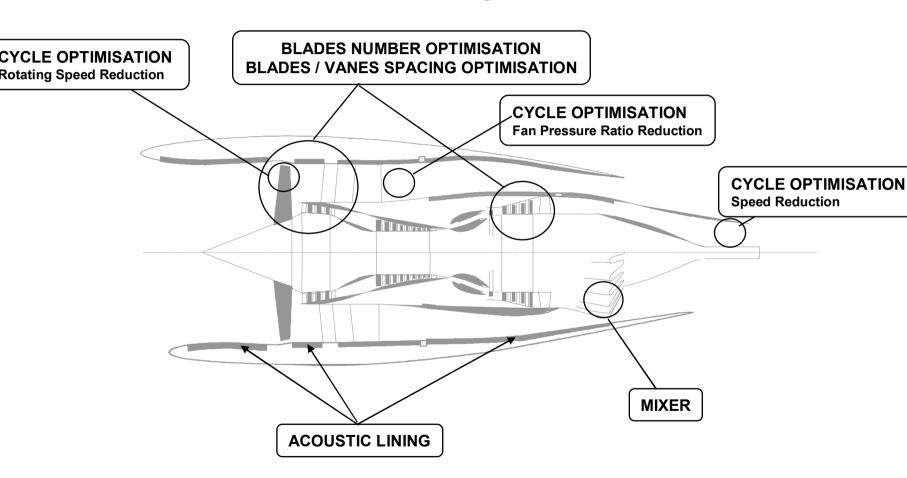
urrent CFMI Engines Typical Noise Signature







urrent Noise Reduction Technologies

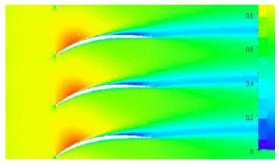






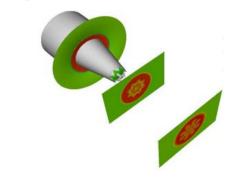
nprovement of Current CFM56 Engines

Fan Noise Reduction through 3D OGV Aeracoustic Design





Jet Noise Reduction through Chevron Nozzle Design



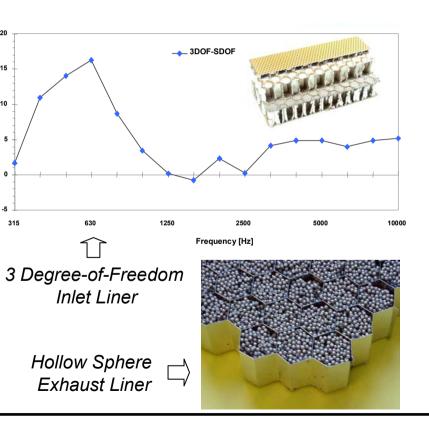




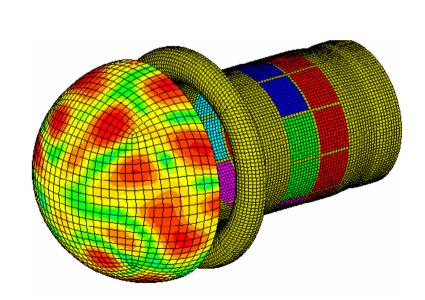


acelle Noise Reduction Systems

Extension of Attenuation Bandwidth towards Low Frequency Range



Improvement of Liner Efficiency through 3D Impedance Optimisation



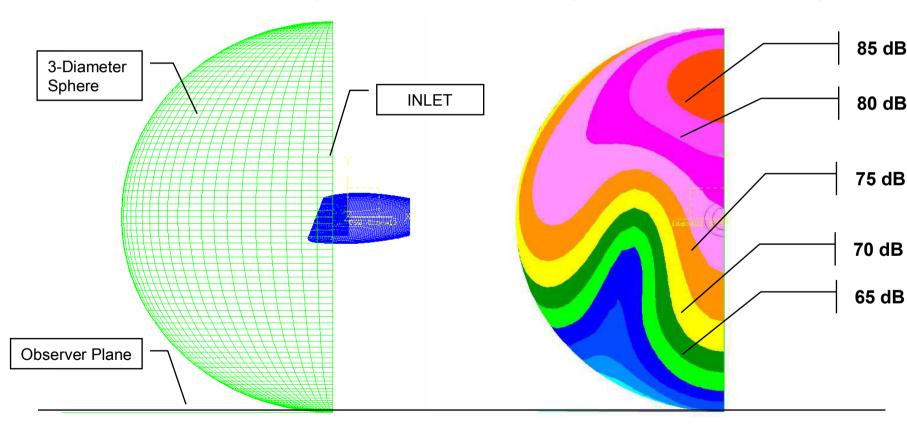




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acelle Noise Reduction Systems

Finite Element Duct Progation Model to support Negatively Scarfed Inlet Design

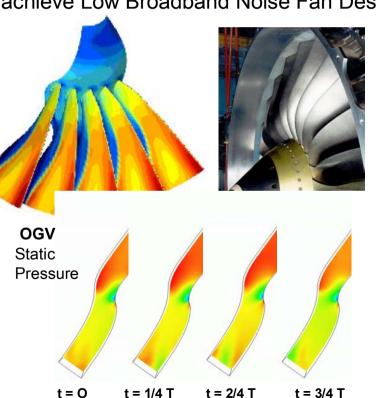




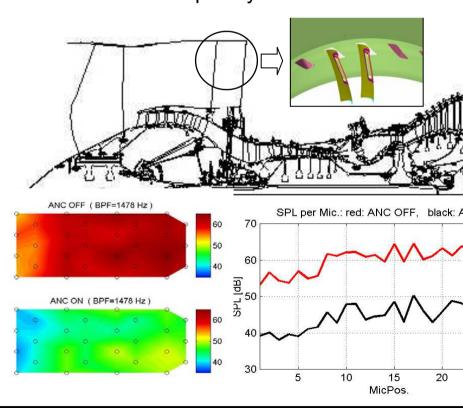


uture Engine Applications

Full Implementation of CAA capabilities achieve Low Broadband Noise Fan Design



Active Technologies applied to Low Frequency Fan Tones Reduction

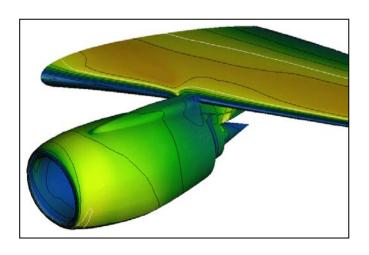






ow Noise Aircraft Design- Engine installation

- Optimisation of future aircraft designs should consider powerplant installation factors as an opportunity for further noise reduction
- Significant development of aerocoustics modelling and appropriate testing facilities will support such activities







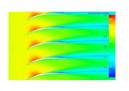
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chnologies Panel to support Optimum Aircraft System Definiti

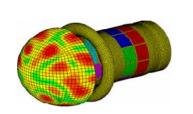
Tools urce anding Advanced CFD Models

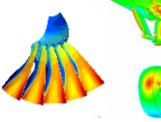
Source Models

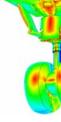
Propagation Models











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Noise Reduction at Source

Noise Reduction Systems





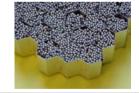




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Nozzle Design & Liner Technology







e noise ction iques

High Lift Devices & Landing Gear













- TO CENTRAL CONTROL OF THE CONTROL OF
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or a Quiet Future





pcoming Challenges

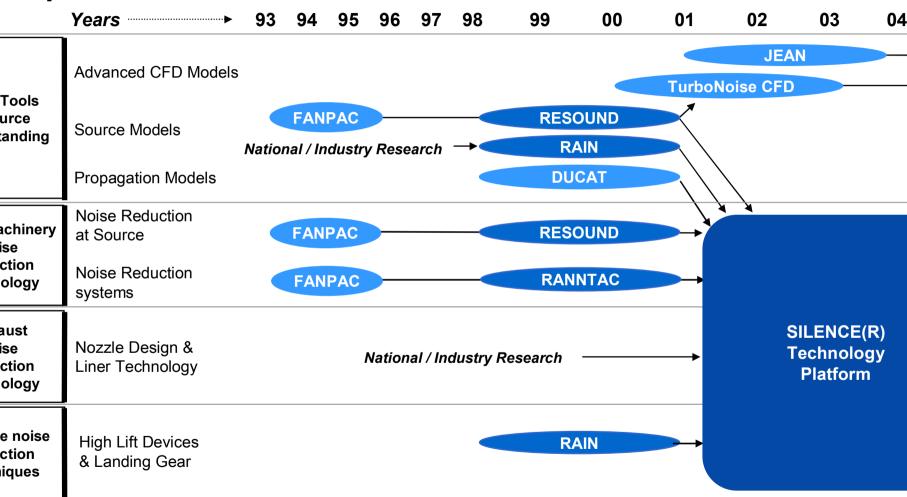
Goal to introduce in service by 2010 products allowing traffic growth at no environmental cost (Ex : 10 dB cumulative margin re Chapter 4 for a typical Tons MTOW Twin Engine Aircraft) :

- ⇒ Development of appropriate panel of noise reduction technologies to support individual optimisation of aircraft system components:
 - Engine
 - Nacelle
 - Landing Gear
 - High Lift Devices
- ⇒ Combined optimisation of powerplant and aircraft taking into consideration installation factors and flight performance

or a Quiet Future



uropean Aircraft Noise Research Initiative



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SILENCE(R) Technology Platform



ct Coordinator : Snecma Moteurs

<u>Innovations</u>:

Evaluation of Advanced Engine Concepts

tion of Novel Noise Reduction Solutions:

Low Noise Engine Component Design

Nacelle and Nozzle Liner Concepts

Active Noise Control Applications

Inlet and Nozzle Advanced Design

ation of Solutions to Helicopter Engine

tions of Ainforms Noise Deduction Color

tions of Airframe Noise Reduction Solutions;

anding Gear

High Lift Devices

ct Duration: 4 years (Start 04/01)

Budget: 112 MEuros (50% EC Support)

ipation: 51 partners from 14 EU countries

+ 2 Associated States.



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